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OXCART WEEKLY REVIEW

8 January - 25 January

	1. Meetings and Contacts:	
25X1A		5X1A
	Fratt & Whitney the hydraulic pump problems connected with the J-58	
	engine.	
25X1A	b. On 9 and 10 January attended the SAE pro-	
*	fessional meetings in Detroit, Michigan.	
÷		
25X1A	c. On 11 January was visited 25	5X1A
20/(1/(by as reported in OXC-2959, dated 16 January 1962.	->/ 4 A
25X1A	d. On 15 January Messrs. Kiefer, Parangosky and of Head-	5X1A
	and the second s	5X1A
25X1A	linear correlator problems in connection with	
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OEV4 A	f. Colonel Beerli,	
25X1A	attended the first LAC OX	5X1A
25X1A	Tamiliarization course at Euroank detween 10 and 10 sanuary.	$\mathcal{M} \sqcup \mathcal{A}$
25X1A	trip report, ONC-2973, dated 22 January 1962, has been forwarded to DD/P.	
	iorwarded to bb/r.	
25X1A	g. Eugene Kiefer was and LAC between 16 and 18 January 25	5X1A
20/(1/(in connection with and the Kirkpatrick visit	
25X1A	172	
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25X1A

	rce C-23 arrived at Stewart Air Force light test program. This aircraft had this mission.
i. Between 15 and 19 Januar	ry
of were TDY LAC where they we	
(1) Firewel is rework	ing the faulty linkage system.
(2) is rework	ing and strengthening the harnesses.
for the drops. It is planned the	will give us a capability of 50,000 feet at tests will be resumed on 29 January o the linkage system thirty drops can be
k. repor	ts that in January he visited Burbank,
	port, OXC-2918 of 22 January, has been
1. Engine test time accumul	lated for the period 1 - 23 January 1962:
Total engine time	73 hours
Afterburner time	2 hours
D-20 engine time	73 hours
Hot inlet time	O
Hot turbine time	1 hour
(1) Four sea level ter FX-111, 113, XD-1, XD-2.	st stands are in operation with engines

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Engine FX-116 (ejector) is mounting in No. 5 see level stand.

for inspection after test completion.

(2) Engine FX-115 (endurance) is mounting in the altitude stand.

(3) Engines FX-112, 114, and 118 have been returned to assembly

(4) Primary effort continues on turbine inlet temperature profile; secondary effort on hydraulic system and controls development.

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OXC-2989 Copy 9 of 9 Page 3 (5) The favorable turbine inlet temperature profile recently established on engine FX-113 has proved inconsistent. 25X1A feels that this is a further indication of diffuser case flow instability, for which initial corrective hardware is now on test with FX-111 and XD-1. This will delay initiation of endurance testing well beyond the three weeks estimated on 4 January. (6) FX-114 teardown inspection has revealed good condition of bolted compressor rotor but damage to most turbine blades and vanes resulting from one burner can mounting bolt failure. (7) The first fully integrated single lever control system using non-production hardware completed 14 hours of non afterburning operation on engine FX-114 without instability. Two combinations of a similar fully integrated single lever system have completed 5 non afterburning hours on engine XD-2 without instability. XD-2, with the latter fully integrated but two lever system, has initiated afterburning operation without instability. The second lever is required to initiate afterburning without fulfilling rated turbine inlet temperature. (8) The first production main fuel and exhaust nozzle controls have been delivered to Florida. 2. Agenda for 26 January Development Branch meeting. 25X1A a. Further Discussion of Engine Status if Required b. Possible Rescheduling of A-12 Trainer - Mr. Kiefer. 25X1A e. Cover Story Status d. Change in Aircraft Engine Out Ferformance - Mr. Kiefer. 25X1A e. Report on the Kirkpatrick Visit - Mr. Parangosky. - 11 Distribution: JOHN PARANGOEKY Copy #1 - DD/P C/DB/DPD 2 - AC/DPD 3 - EXO/DPD 4 - ASST C/DPD

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